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# Multi-objective optimization on dimple shapes for gas face seals

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## ABSTRACT

The shape of surface texture has a significant effect on the performance of face seals. Many studies found that the textures designed for improving load carrying capacity tend to increase leakage at the same time. Therefore, a multi-objective optimization approach is presented to optimize the dimple shapes with freedom edges. It has been found the dimples with asymmetric "V" shape offer better performance in terms of load carrying capacity and leakage. Moreover, the optimal shapes are compared with four kinds of optimal regular shapes under different rotating speeds. The results show that the superiority of shape optimization is more obvious in conditions with high speed.

## 1. Introduction

There is an increasing demand for reliable and durable face seals with low leakage and low friction under high speed, high temperature, and complex working conditions. Surface texturing has been proven to be an effective means to improve the tribological performance of sliding surfaces because of the hydrodynamic effect under full or mixed lubrication and lubricant reservoir effect under staved lubrication conditions [1–5].

The patterns of grooves and dimples are common types of surface texture for face seals [1-15]. They are typical representatives of connected texture and disconnected texture, respectively. Grooves such as spiral grooves [10-13] and T-shape grooves [9] have been widely used in face seals due to the pumping effect and hydrodynamic effect. Dimples were also proven to reduce friction or improve load carrying capacity under different conditions [3-7,14-18], meanwhile, they were expected to obtain a better hydrodynamic effect due to the disconnected structure [7,15]. Nakano et al. [7] found that cast iron surfaces with dimples had lower friction than the surfaces with grooves or meshes under lubricated conditions. Shi et al. [14] found that elliptical dimples with a high area density can obtain a higher load carrying capacity and higher gas film stiffness than grooves for gas lubrication. With further research, dimple shape has become a hot factor in the optimization of surface textures. Lu and Khonsari [2] reported that bushing with elliptical dimples has a lower friction coefficient than that with circular dimples under mixed lubrication. Uddin et al. [19] found that square shape dimples with a single wedge-bottom profile offered better tribological performance than triangular, chevron, circular and

elliptical shapes. In view of the above studies, a conclusion can be drawn that the dimple shape is an essential factor to improve the tribological performance. Therefore, beyond the above regular shapes, are there any other shapes of dimples which can provide a better impact on the performance of sliding surface? Shen and Khonsari [20,21] conducted numerical optimization for complex dimple shapes using a sequential quadratic programming (SQP) algorithm. The optimal shape which can produce the maximum load carrying capacity was obtained by changing the design variables from an arbitrary shape. This work provides a possibility for further optimization of texture shapes.

Leakage is another important factor for gas face seals. However, it is difficult to improve the load carrying capacity or opening force and simultaneously reduce the leakage through surface texturing [8,10,22]. The clearance of seal rings increases with the increasing opening force, which leads to an undesirable increase of leakage [8]. In order to ease the case, the seal with double-row spiral grooves was presented where one row can pump the leaked medium back to the sealed space. But such a seal tends to have a complicated structure, needs larger installation space and can only be applied under low pressure differences [8]. Moreover, the load carrying capacity and the leakage rate were analyzed independently in most of the studies. It is difficult to obtain a better combination performance objectively by a single objective analysis.

This study aims to provide a multi-objective optimization approach specially for conflicting objectives i.e., load carrying capacity and leakage rate, to optimize the shape of dimples on gas face seals. The models of dimples and multi-objective optimization problem are established where the dimples have an arbitrary shape on a certain

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Nomenclature		Т	thermodynamic temperature
		$r_{\mathrm{I}}$	inner diameter of sealing ring
h	gas film thickness	ω	angular velocity
$F_t$	non-dominated front	r <sub>o</sub>	outer diameter of sealing ring
$h_0$	sealing clearance	W	load carrying capacity
Μ	average molar mass of gas	α	small angle of dimple region of type b
$h_{g}$	groove depth	$\overline{R}$	dimensionless coordinate in the radial direction
$P_0$	initial population	β	large angle of dimple region of type b
1	inner diameter of dimple region of type a	9	dimensionless coordinate in the circumferential direction
$P_t$	population at N <sub>th</sub> iteration	Θ	coordinate in the circumferential direction
т	number of design variables	$\overline{H}$	dimensionless gas film thickness
Q	leakage rate	γ	gas pressure to density ratio
n <sub>r</sub>	rotating speed	$\overline{P}$	dimensionless gas film pressure
$Q_t$	population obtained by crossover and mutation	δ	angle of computing domain
р	gas film pressure	$\overline{Q}$	dimensionless leakage rate
R	gas constant	ρ	gas density
$p_a$	atmospheric pressure	$\overline{W}$	dimensionless load carrying capacity
$R_t$	union of $P_t$ and $Q_t$	μ	dynamic viscosity of gas
r	coordinate in the radial direction	NSGA-II	elitist non-dominated sorting genetic algorithm

constraint. Then, the optimal dimple shapes and the Pareto-optimal sets are obtained using the elitist non-dominated sorting genetic algorithm (NSGA-II). Furthermore, the optimal shapes are compared with the optimal regular shapes, including circle, ellipse, square and triangle, under different rotating speeds.

# 2. Optimization method

## 2.1. Physical model and governing equation

The face seal is composed of two sealing rings. Fig. 1 shows the physical model, where  $r_1$  and  $r_0$  are the inner radius and outer radius of the rings, respectively. The sealing faces are separated by a layer of gas film with the thickness of  $h_0$ .  $n_r$  is the rotating speed, and  $h_g$  is the depth of dimples. In this study, 24 dimples are uniformly distributed on the stationary ring face. A sector unit cell containing one dimple is considered as the computing domain, and  $\delta$  is the angle of computational domain.

In order to optimize the shape of the dimple from arbitrary geometry, there are 2n control points for one dimple which is formed by connecting the adjacent points. In this study, the shapes of dimples include two types: type *a* and type *b*, as shown in Fig. 2. For type *a*, the control points of *n*-*i*<sup>th</sup> and 2n-*i*<sup>th</sup> (i = 0, 1, ..., n-1) are located at the same radius, and they are free in the circumferential direction under a certain constraint which is shown in the multi-objective optimization model. The coordinates of these points are shown in Table 1. Based on this principle, 2n + 2 design variables are required, and they are  $\theta_1, \theta_2, ..., \theta_n$ ,  $\theta_1^a, \theta_2^a, ..., \theta_n^a, l, L$ . For type *b*, the regularity is similar to that of type *a*, but the points of *n*-*i*<sup>th</sup> and 2n-*i*<sup>th</sup> (i = 0, 1, ..., n-1) are located at the same angle and they are free in the radial direction.

The number of design variables m = 2n+2 is an important factor for the computing accuracy and efficiency. Generally, the more the design variables, the smoother the edges of optimal shapes, however, the longer the computing time. Take type *a* as an example, the number of design variables *m* is studied in this study. It is found that when *m* is increased from 8 to 16, the optimal shapes have similar geometries and the edges become smoother and smoother, meanwhile, the computing times increase at an increasing rate. For m = 14, the optimized shapes have relatively smooth edges, and the computing time is 247 h which is 32% smaller than the computing time (367 h) of m = 16. Based on the computing time and shape accuracy, m = 14 is chosen in this study.

The two-dimensional steady-state Reynolds equation in the polar coordinates is employed to analyze the gas film pressure distribution p (r, $\theta$ ). The equation can be expressed as:

$$\frac{\partial}{\partial r} \left( \frac{\rho r h^3}{\mu} \frac{\partial p}{\partial r} \right) + \frac{1}{r} \frac{\partial}{\partial \theta} \left( \frac{\rho h^3}{\mu} \frac{\partial p}{\partial \theta} \right) = 6\omega r \frac{\partial(\rho h)}{\partial \theta}$$
(1)

where *r* and  $\theta$  are the polar coordinates in the radial and circumferential directions, respectively;  $\mu$  is the dynamic viscosity of gas, which is assumed as a constant in this study;  $\rho$  is the gas density; *p* is the gas pressure; *h* is the gas film thickness which is equal to  $h_0 + h_g$  in the dimple region and equal to  $h_0$  beyond the dimple region; and  $\omega$  is the angular velocity which is equal to  $2\pi n_r/60$ .

Reynolds number, defined as  $Re = \frac{\rho vh}{\mu}$  where v is the peripheral velocity of the seal face, is a dimensionless number that characterizes the state of fluid. Generally, the flow can be treated as laminar when Re < 1000 [23]. In this study, an example with the rotating speed of 10000 rpm is mainly studied, where the maximum Re is not more than 350, so laminar flow is assumed in this model.



Fig. 1. Physical model of the face seal.



Fig. 2. Geometrical models of dimples in polar coordinate.

 Table 1

 Coordinates of the points in the physical model of dimples.

	Points	$i (i = 1, 2 \dots n)$	$j (j = n+1, n+2 \dots 2n)$
Coordinates	Type $a (dr = (L-l)/(n-1))$ Type $b (d\theta = (\beta-\alpha)/(n-1))$	$(l + dr \times (i-1), \theta_i)$ $(r_i, \alpha + d\theta \times (i-1))$	$ \begin{array}{l} (l + dr \times (j\text{-}n\text{-}1), \ \theta_{j\text{-}} \\ {}_{n} + \theta_{j\text{-}n}^{a}) \\ (r_{j\text{-}n} + r_{j\text{-}n}^{b}, \ \alpha + d\theta \times (j\text{-}n\text{-}1)) \end{array} $

It is assumed that the sealing gas is of uniform temperature, because the friction of non-contact gas seals is quite small and the temperature rises slowly. Moreover, it is considered as an ideal gas. The ideal gas state equation is written as:

$$\frac{p}{\rho} = \frac{RT}{M} = \gamma \tag{2}$$

where *M* is the average molar mass of gas, equal to 29 g/mol; *R* is the gas constant, equal to  $8.314 \text{ cm}^3 \text{MPa/mol}$ ·K; *T* is the absolute temperature of the sealing system and is taken as 300 K in this study. The ratio of *p* and  $\rho$  is defined as  $\gamma$  and will be used in the calculation of the leakage rate.

The pressures at the outer radius and inner radius are set to  $p_{\rm I}$  and  $p_{\rm O}$ , respectively, and the periodic boundary condition is applied in the circumferential directions to account for the interaction between dimples. The boundary condition is written as:

$$\begin{cases} p(r = r_{\rm i}, \theta) = p_{\rm I} \\ p(r = r_{\rm O}, \theta) = p_{\rm O} \\ p(r, \theta = 0) = p(r, \theta = \delta) \\ \frac{\partial p}{\partial \theta}(r, \theta = 0) = \frac{\partial p}{\partial \theta}(r, \theta = \delta) \end{cases}$$
(3)

The performance parameters, including load carrying capacity W and leakage rate Q, are expressed as:

$$W = \int_0^\delta \int_{r_1}^{r_0} p(r,\theta) r dr d\theta$$
(4)

$$Q = -\int_{0}^{\delta} \left(\frac{prh^{3}}{12\mu\gamma}\frac{\partial p}{\partial r}\right)_{r=r_{1}} d\theta$$
(5)

The load carrying capacity W is the integral of the gas film pressure across the entire computing domain. The leakage rate Q is the flow rate in the radial direction and it follows the flow continuity principle. It could be calculated no matter which radius is chosen. So the inner radius is adopted in the following evaluation.

The equations and performance parameters are nondimensionalized based on the dimensionless terms:

$$\overline{R} = \frac{r}{r_1}, \ \overline{H} = \frac{h}{h_0}, \ \overline{P} = \frac{p}{p_a}, \ \overline{W} = \frac{W}{p_a r_1^2}, \ \overline{Q} = \frac{12Q\mu\gamma}{p_a^2 h_0^3}$$
(6)

The dimensionless Reynolds equation can be expressed as:

$$\frac{1}{\overline{R}}\frac{\partial}{\partial \overline{R}}\left(\overline{RPH}^{3}\frac{\partial\overline{P}}{\partial \overline{R}}\right) + \frac{1}{\overline{R}^{2}}\frac{\partial}{\partial \theta}\left(\overline{PH}^{3}\frac{\partial\overline{P}}{\partial \theta}\right) = \Lambda \frac{\partial(\overline{PH})}{\partial \theta}$$
(7)

where  $\Lambda = \frac{6\mu\omega r_1^2}{p_a h_0^2}$  is the compressibility number. The successive over relaxation (SOR) method is utilized to solve the

The successive over relaxation (SOR) method is utilized to solve the dimensionless Reynolds equations. The dimensionless performance parameters are obtained according to Eq. (4), Eq. (5) and Eq. (6).

## 2.2. Multi-objective optimization model

For a multi-objective optimization problem, usually there is not a single optimal solution but an optimal set called as Pareto-optimal set. The element in Pareto-optimal set is called as Pareto-optimal solution which means that one objective is the best when other objectives are fixed. In other words, it is impossible to make one of the objectives better than the optimal solution without destroying any other objectives. The multi-objective optimization model can be expressed as:

min 
$$f(x) = [f_1(x), f_2(x), \dots, f_m(x)]$$
 subject to (s.t.) x  
meets certain constraints (8)

where  $f_1(x)$ ,  $f_2(x)$ , ...,  $f_m(x)$  are the objective functions, and x is a vector formed by independent variables.

In this study, the dimensionless load carrying capacity  $\overline{W}$  and the dimensionless leakage rate  $\overline{Q}$  are the objectives, and the dimensionless design variables related to the dimple shapes are the independent variables. Because the minimum values of objective functions are the goal of the multi-objective optimization,  $-\overline{W}$  is taken as an objective according to the practical meaning of  $\overline{W}$ . The model of multi-objective optimization for dimples of type *a* is expressed as:

where  $x_a$  is a vector formed by the dimensionless parameters of dimples of type *a*.

Similarly, the model of multi-objective optimization for dimples of type *b* is expressed as:

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where  $\overline{r}_i = r_i/r_i$ ,  $\overline{r}_i^b = r_i^b/r_i$  (*i* = 1,2,...,*n*), and  $x_b$  is a vector formed by the dimensionless parameters of dimples of type *b*.

The multi-objective optimization problem is solved using the elitist non-dominated sorting genetic algorithm (NSGA-II). The algorithm brings forward a fast non-dominate sorting approach; and its computing complexity is reduced greatly [24,25]. Fig. 3 presents a flowchart of NSGA-II. The optimal results are obtained through repeated iteration. Staring from a random population  $P_0$ , a new population  $Q_t$  is obtained by genetic manipulation for the population  $P_t$ , then, a non-dominated front  $F_t$  is formed by sorting and selecting for the union set  $R_t$  of  $P_t$  and  $Q_t$ . The population  $P_{t+1}$  is formed by individuals in  $R_t$  corresponding to the top N individuals in  $F_b$  where N is the population size. The work repeats until the termination condition that the population distance of adjacent iterations is less than  $10^{-5}$  is satisfied.

The values of geometric parameters and condition parameters used in this study are shown in Table 2.

#### 3. Results and discussion

### 3.1. Shape optimization results

The multi-objective optimization problem is solved using NSGA-II according to the flowchart in Fig. 3. The Pareto-optimal sets and optimal objectives for type a and type b are obtained, respectively. The number of solutions in Pareto-optimal set is related to the population size and Pareto fraction. Too many solutions are unnecessary because they are continuous and the adjacent solutions have high similarities.



Fig. 3. Flowchart of the NSGA-II.

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Table 2

Geometric parameters and condition parameters.

Items	Values
Inner radius of computing domain $r_{\rm I}$ , mm	17.9
Outer radius of computing domain $r_0$ , mm	23.4
Angle of computing domain $\omega$	π/12
Depth of dimples $h_g$ , µm	10
Gap of two specimen $h_0$ , µm	10
Sealed gas dynamic viscosity $\mu$ , MPa·s	$18.448 \times 10^{-12}$
Atmospheric pressure $p_a$ , MPa	0.101
Boundary pressure of inner side $p_{I}$ , MPa	0.202
Boundary pressure of outer side $p_0$ , MPa	0.101
Rotating speed $n_r$ , rpm	3000, 10000, 20000



Fig. 4. The optimal solutions of multi-objective optimization.

So it is set to 7 in order to show the optimal shapes clearly.

Fig. 4 shows the solutions of the multi-objective optimization for type a and type b. The horizontal axis is the dimensionless leakage rate and the vertical axis is the dimensionless load carrying capacity. Take type a as an example, a curve is formed by connecting the optimal points, which divides the area into two parts. The optimal solutions are the boundary of the shadow area. For all dimple shapes of type *a* except optimal shapes, their dimensionless load carrying capacities and dimensionless leakage rates will fall into the black shadow area. Similar conclusion can be drawn for type b. As shown in Fig. 4, the growth rate of optimal dimensionless load carrying capacity is getting smaller as the optimal dimensionless leakage rate increases linearly. In other words, the optimal dimensionless leakage rate will increase sharply when the optimal dimensionless load carrying capacity increases slightly. That is to say, blindly pursuing the maximum load carrying capacity is not appropriate in the design of dimples for gas face seals because drastic rises may occur in the leakage rate.

As can also be observed in Fig. 4, the optimal dimensionless load carrying capacity of type *b* is smaller than that of type *a* under the same dimensionless leakage rate for all solutions, meanwhile, the optimal dimensionless leakage rate of type *b* is larger than that of type *a* under the same dimensionless load carrying capacity. For example, the optimal dimensionless load carrying capacity of type *a* is 2.52 when the dimensionless leakage rate is 57.9, higher than that of other type *a* with the same dimensionless leakage rate. It is 2.18 for type *b*, which is 13.5% lower than that of type *a*. Similarly, the optimal dimensionless leakage rate of type *a*. Similarly, the optimal dimensionless load carrying capacity is 57.9 for type *b*, which is 23.7% higher than that of type *a*. In view of this, the optimal shapes of type *a* can provide a better combination performance than that of type *b*. The reason may be that the freedom in the circumferential direction is beneficial to produce a larger load carrying capacity or a lower leakage



Note: the black line is the profile of dimple and the contour is the dimensionless pressure distribution



Table 3

Fig. 5. Mechanism of optimal shape of type a.

rate, but it was constrained for the shape of type *b*.

Moreover, as can be seen in Fig. 4, the value spans of optimal objectives are wide. That is to say, the optimization results can provide a reference for a variety of leakage or load carrying capacity requirements, meanwhile, the objectives span of type a is larger than that of type b. For example, when the dimensionless leakage rate of 74.6 is allowed, the optimal solutions of type b can't provide a more effective reference for the design of dimples. The optimal shape of type a corresponding to the dimensionless load carrying capacity of 2.73 can be selected. In this way, the dimples can satisfy the requirement of leakage rate and can generate a load carrying capacity as large as possible at the same time. Considering this view and the above analysis, the optimal shapes of type a are better than those of type b.

Table 3 shows the optimal dimple shapes and the dimensionless pressure distributions corresponding to the above solutions. The detailed values of the design variables are shown in Appendix. As shown in Table 3, the optimal shapes for type *a* have very similar geometries with asymmetric "V" apart from the solution 1. The widths and areas of edges near inner side are larger than those near outer side. As the solution order increases, the widths and areas of edges near inner side increase but the edges near outer side are almost same. It is hard to get such special optimal shapes through imagination or subjective designing. Numerical optimization may be the only way. As also can be seen in Table 3, the optimal shapes for type *b* are constrained greatly in the circumferential direction, their edges mainly show a jagged shape in the radial direction. But the trend in the radial direction is not a very effective factor according to the optimal solution in Fig. 4. In addition, for type *a* and type *b*, all of the optimal dimples are located at the outer side of sealing faces. The reason maybe that gas flows into the sealing clearance from inner side driven by the high pressure at inner radius, then the gas pressure decreases from inner to outer, and it is further enhanced by the dimples near outer side.

Moreover, as shown in Table 3, the dimensionless pressure distributions between type a and type b are different. For type a, high pressure is mainly generated at the end of wide edge, which is located in the middle of the computational domain. However, there are obvious breaks for the pressure distribution of optimal type b. In general, the optimal shapes of type a have a more uniform pressure distribution than those of type b, which may lead to a higher load carrying capacity. Meanwhile, for most optimal shapes of type a, the dimple areas are smaller than those of type b, which may lead to a lower leakage rate, because the untextured area is a critical factor for the leakage. In addition, there are two flows including pressure flow and shear flow in the computational domain for the optimal shape of type a, as shown in Fig. 5. High pressure is also generated at the end of the narrow edge, and the mediums will flow back to inner side driven by the pressure difference. So the leakage rate will be reduced to some extent.

### 3.2. Comparison with optimal regular shapes

## 3.2.1. Regular shapes

The optimal dimple shapes of type a are compared with optimal regular shapes including circle, ellipse, square and triangle. The design variables are determined according to their features. Table 4 shows the regular shapes and their design variables. Different constraints are applied to the design variables for different dimples to ensure that the dimples do not exceed the computational domain. For all of these dimples, their depths are equal to the dimple depths of type a. The optimal solutions are obtained using the same multi-objective optimization method.

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No.	Description	dimple shapes	Design variables	Number of design variables
1	Circle	le re	α <sub>c</sub> , l <sub>c</sub> , r <sub>c</sub>	3
2	Ellipse	la le le actual de la competencia de la competen	$a_{e1}, l_{e1}, a_{e2}, l_{e2}, l_{e2}, l_{e}$	5
3	Square		$\alpha_{s1}, l_{s1}, \alpha_{s2}, l_{s2}$	4
4	Triangle	10 10 10 10 10 10 10	$\alpha_{t1}, l_{t1}, \alpha_{t2}, l_{t2}, \\ \alpha_{t3}, l_{t3}$	6



Fig. 6. Optimization results for different dimple shapes.

#### 3.2.2. Optimal results for regular shapes

Fig. 6 shows optimal solutions of type *a* and regular shapes. Table 5 shows the corresponding optimal regular shapes and their dimensionless pressure distributions. Similar to Fig. 4, the growth rate of optimal dimensionless load carrying capacity is getting smaller as the optimal dimensionless leakage rate increases. In addition, for all shapes except optimal shapes, their dimensionless load carrying capacity and dimensionless leakage rate will fall below the optimal curves.

It can also be seen in Fig. 6 that the optimal dimensionless load carrying capacities of regular shapes are smaller than those of type a under the same leakage rate in most cases. That is to say, the optimal shapes of type a can offer better performance in comparison with other optimal regular shapes. For example, the optimal shape of type a has

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the dimensionless load carrying capacities of 1.47 and 2.67 when the dimensionless leakage rates are 30 and 70, which is 23.5% and 16.1% higher than the lowest values of 1.19 and 2.30, respectively. Meanwhile, the optimal shape of square has the second largest optimal dimensionless load carrying capacity 2.44 when the dimensionless leakage rate is 70, but it has the lowest optimal dimensionless load carrying capacity 1.19 when the dimensionless leakage rate is 30. Generally speaking, the optimal shape of type *a* is the most satisfactory under both low and high leakage rate conditions, and the optimal square shape can be the second choice under a high leakage rate condition. Certainly, the optimal square shapes of type *a* under the same leakage rate, but they have a lower design complexity.

The wide spans of optimal objectives mean that the optimization results can provide references for a variety of leakage or load carrying capacity requirements. In view of this, the optimal shapes of type *a* and optimal square shapes are more favorable than other optimal shapes, because they have larger spans.

As shown in Table 5, similar to the optimal shapes of type *a*, as the solution order increases, the areas of optimal regular shapes are becoming larger in general, and the locations of dimples move to the upstream and outer side. For the optimal square shapes, the increasing length in the radial direction plays an important role in preventing flow, improving film pressure and enlarging the leakage rate at the same time. The decreasing length in the circumferential direction can control the leakage rate to some extent. So the length-width ratio, defined as the ratio of the length in the circumferential direction to the length in the radial direction of the square dimple, is a sensitive factor for the pressure distribution.

In summary, the optimal shapes of type a are the most satisfactory. The optimal square shapes can be the second choice because of the large span of optimal solutions and the simple structures.

#### Table 5

Optimal regular shapes and dimensionless pressure distributions.



#### Table 6

Optimal shapes of type a and dimensionless pressure distributions for  $n_r = 3000$  rpm and  $n_r = 20000$  rpm.





Fig. 7. Optimal solutions of different dimple shapes: (a)  $n_r = 3000$  rpm, (b)  $n_r = 20000$ .

## 3.2.3. Results under different rotating speeds

In order to verify if the rotating speed has obvious influence on the optimal shapes, the same approach is carried out for a quite low speed  $n_r = 3000$  rpm and a quite high speed  $n_r = 20000$  rpm under the same laminar assumption, although the maximum *Re* is about 1675 for the speed of 20000 rpm, and turbulence flow should be considered in real case.

Table 6 shows the optimal dimple shapes of type *a* for  $n_r = 3000$  rpm and  $n_r = 20000$  rpm. Although there are small differences between the same order solutions for different speeds, in general, they are similar.

Fig. 7 shows the optimal solutions of different dimple shapes for  $n_r = 3000$  rpm and  $n_r = 20000$  rpm. The optimal shape of type *a* shows a higher dimensionless load carrying capacity than other optimal shapes under the same dimensionless leakage rate for both low speed and high speed. Similarly, it can get a lower dimensionless leakage rate than other shapes under the same dimensionless load carrying capacity. Meanwhile, the solution spans of optimal type *a* are larger than other shapes, which means a broad guidance on different leakage or load carrying capacity requirements. Moreover, as can be seen in Figs. 6 and 7, the differences of optimal solutions between optimal shape of type *a* and other optimal shapes increase with the increasing rotating speed. The advantage of optimal type *a* is more obvious at high rotating speed.

The specific samples show the effectiveness of the multi-objective optimization for the dimple shape, and other working conditions can also be analyzed using this method when the actual condition

# Appendix

Table .	A1
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Detailed values of the optimal shapes of type *a* in Table 3

parameters are specified.

### 4. Conclusions

In this study, a multi-objective optimization approach is presented to optimize the dimple shape for gas face seals.

For any types of dimple shape, each optimal shape can get a highest load carrying capacity under a given leakage rate or a lowest leakage rate under a given load carrying capacity. The wide spans of solutions mean the results can provide a reference for a variety of leakage or load carrying capacity requirements.

The optimal shapes of asymmetric "V" always show a best impact on the performance of gas face seals. The advantage of shape optimization using multi-objective approach for gas face seals is more obvious at high speed conditions.

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Solutions orders	1	2	3	4	5	6	7
$ heta_1$	0.25909	0.25559	0.25192	0.23943	0.22239	0.18404	0.17701
$\theta_2$	0.00159	0.00458	0.06449	0.15392	0.12496	0.12429	0.1204
$\theta_3$	0.00193	0.00318	0.01687	0.05402	0.10313	0.11217	0.11474
$ heta_4$	0.00083	0.12445	0.1197	0.11791	0.11008	0.11235	0.11432
$\theta_5$	0.25886	0.22637	0.20278	0.19645	0.21226	0.21676	0.23036
$ heta_6$	0.00158	0.00306	0.0138	0.02404	0.02054	0.02008	0.01266
$\theta_1{}^a$	0.00101	0.00147	0.00466	0.011	0.01633	0.0177	0.02308
$\theta_2{}^a$	0.00099	0.00721	0.01778	0.07536	0.12821	0.13002	0.1382
$\theta_3{}^a$	0.2361	0.24411	0.24036	0.19893	0.15361	0.15056	0.14794
$\theta_4{}^a$	0.0033	0.11499	0.12739	0.13742	0.14145	0.14516	0.14768

$\theta_5{}^a$	0.00124	0.01964	0.04189	0.06085	0.03942	0.03919	0.02787
$\theta_6{}^a$	0.13152	0.23424	0.23526	0.22722	0.23294	0.23543	0.23637
$l/r_{\rm I}$	1.00469	1.12325	1.14479	1.13878	1.13001	1.10213	1.08318
$L/r_{\rm I}$	1.17265	1.27475	1.27935	1.2817	1.2835	1.28419	1.28461

Table A2 Detailed values of the optimal shapes of type b in Table 3

Solution orders	1	2	3	4	5	6	7
<b></b> <i>r</i> <sub>1</sub>	1.12845	1.12189	1.11975	1.10649	1.08297	1.07185	1.05829
r <sub>2</sub> <del>r</del> <sub>3</sub>	1.12169	1.12526	1.11519	1.13936	1.08934	1.06188	1.03061
r <sub>4</sub> r <sub>5</sub>	1.12183	1.12/53	1.12299	1.14132	1.11337	1.13477	1.13377
<b></b> <i>F</i> <sub>6</sub> <i>F</i> <sub>1</sub>	$1.11198 \\ 0.15832$	1.07776 0.16872	1.07559 0.17237	1.15367 0.19568	1.13768 0.21298	1.09672 0.21882	1.08345 0.22547
₽ <sub>2</sub> ₽ <sub>3</sub>	0.17237 0.13401	0.16164 0.10485	0.15569 0.09326	0.16070 0.05913	0.16944 0.02294	0.18174 0.02272	0.20927 0.02267
<u>₹</u> 4 ₹5	0.17471 0.15231	0.16803 0.12844	0.16546 0.05070	0.16591 0.09884	0.16960 0.13852	$0.17808 \\ 0.15376$	0.17517 0.15622
$\overline{r}_{6}$	0.16883 0.11970	0.13301 0.01260	0.11488 0.01300	0.13268 0.01041	0.13934 0.01286	0.17770 0.01480	0.18006 0.01495
β	0.26018	0.25945	0.25893	0.23485	0.21948	0.20462	0.20174

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